

PREDICTION OF FATIGUE STRENGTH OF SHOT-PEENED AND THEN GROUND SPECIMENS

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ABSTRACT

In this paper a new concept of shot-peening strengthening, internal fatigue strength, is advanced. According to the internal fatigue strength of the target material, a method for prediction of fatigue strength of the peened and then ground specimens is proposed and described in detail. The method mainly includes calculation of the compressive and tensile residual stress field, determination of the internal fatigue strength, and prediction of the nominal fatigue strength of the peened specimen. Experimental verification is carried out with 40 Cr steel specimens under different heat treating and peening conditions. The results of verification show that the prediction method is both simple and accurate.

KEYWORDS

Shot peening, Residual stress, Fatigue strength

NOMENCLATURE

σ_{max}	Maximum resultant tensile stress
σ_c, σ_t	Compressive and Tensile residual stress
σ_{cs}	Compressive residual stress at surface
σ_{cm}	Maximum compressive residual stress
σ_{tm}	Maximum tensile residual stress
$\sigma_{l(cm)}$	Applied stress at the point where σ_{cm} occurs
FS	Fatigue strength for a $\times 10^6$ -cycle life under three-point bending with stress ratio of 0.05
IFS	Internal fatigue strength
SFS	Surface fatigue strength, i.e., fatigue strength of target material under as-heat treated condition
Z_0	Thickness of compressive residual stress layer introduced during shot peening
Z_{cm}	Depth of point where σ_{cm} occurs
Z_{tm}	Depth of point where σ_{tm} occurs
Z_s	Depth of point where fatigue source occurs
σ_s, σ_b	Yielding and Ultimate tensile strength of target material
D, D_d	Average diameter of steel shots and peening dents
C	peening coverage rate ($\times 100\%$)
h	Thickness of specimen
α	Ratio of IFS and SFS
F	Equivalent load during peening

INTRODUCTION

It has been noticed that when the surface of a shot peened specimen is ground a little or the peening intensity is lower, the crack initiation location during three point bending fatigue test is nearly always in the inside of the specimen and within the tensile residual stress zone [1,2]. The critical fatigue stress of material in the interior, i.e., internal fatigue strength (IFS) is enormously higher than the fatigue strength of un-peened specimen, i.e., surface fatigue strength (SFS). IFS, as being proposed by the authors [3,4], is an intrinsic property of material and is independent of shot peening regime. Experimental and calculated data showed that the ratio of IFS and SFS is about equal to 1.35. IFS determined by the ratio and SFS of a material can be used to predict the apparent or nominal fatigue strength (FS) of peened and the ground specimens. In this paper a method for prediction of fatigue strength in such case is proposed and the predicted results are verified by experiments.

PREDICTION METHOD

Main Train of Thought

As mentioned above, fatigue failures of peened and then ground specimens nearly always start in the tensile residual stress zone. Because of the statistical nature of fatigue, the exact position of fatigue source can not be determined before testing. But from safety consideration, it is reasonable to assume that the failure will begin at the position where maximum resultant tensile stress σ_{max} (which is equal to the sum of local applied stress and local tensile residual stress there) occurs and reaches the internal fatigue strength of target material. Then, if the gradient of applied stress is very small, the criterion of fatigue failure for peened and then ground specimens can be expressed as

$$\sigma_{max} = \sigma_{tm} + \sigma_{l(tm)} = IFS \quad (1)$$

where σ_{tm} is the peak value of tensile residual stress, and $\sigma_{l(tm)}$ is the applied stress at the point where σ_{tm} occurs. According to Eq.(1), if the IFS of target material and σ_{tm} under given peening condition are known, the $\sigma_{l(tm)}$ and then the nominal fatigue strength of specimen, FS, can be calculated

$$\sigma_{l(tm)} = IFS - \sigma_{tm} \quad (1-1)$$

$$FS = (h / (h - 2Z_{tm})) \sigma_{l(tm)} \quad (2)$$

where h is the thickness of specimen, and Z_{tm} is the depth of point where σ_{tm} occurs.

Calculation of Compressive Residual Stress Field

According to a model proposed by the authors[3,4], in order to calculate the σ_{tm} in Eq.(1-1), the compressive residual stress distribution introduced by shot peening should be determined firstly. Two methods for calculation of compressive residual field have been put forward.

Empirical Method. According to authors' work[3], some characteristic parameters of the compressive residual stress field (Fig.1) can be calculated by using following empirical equations.

The compressive residual stress at surface

$$\sigma_{cs} = 114 + 0.563 \sigma_s \quad (\text{in Mpa}) \quad (3)$$

The maximum value of compressive residual stress

$$\sigma_{cm} = 147 + 0.567 \sigma_b \quad (\text{in Mpa}) \quad (4)$$

The thickness of compressive residual stress layer

$$Z_0 = (1.41D_4 - 0.09D)[1 + 0.09(C - 1)^{0.35}] \quad (\text{in } \mu\text{m}) \quad (5)$$

The depth of point where σ_{cm} occurs

$$Z_{cm} \approx 0.28Z_0 \quad (\text{in } \mu\text{m}) \quad (6)$$

In these equations

σ_s and σ_b are yielding and ultimate tensile strength of target material (in Mpa);
 D is median diameter of steel shots (in μm);

D_d is median peening-dent diameter (in μm), which can be obtained by using tentative peening [3,5]; and

C is peening coverage rate ($\times 100\%$).

Knowing these parameters, we can determine the distribution of compressive residual stress in the surface layer rather accurately.

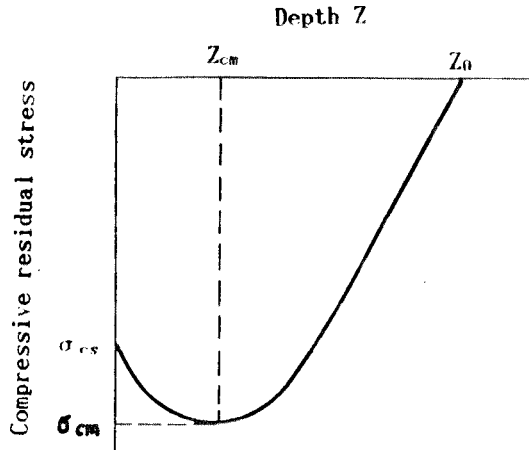


Fig.1 Definition of characteristic parameters of compressive residual stress field

Analytical Method. According to an analytical model proposed by authors [3], the compressive residual stress distribution with coverage rate of 100% can be determined by using a computer program, if the equivalent load (F) during peening and the tension stress-strain curve of target material are known. F can be calculated from median diameter of shots and peening dents (D and D_d) and true stress of target material at necking (S_b), as described in Ref [5]. Peening coverage rate (C) has little influence on other parameters but increases the thickness of the compressive residual stress layer

$$Z_0 = Z_{0(c=1)} [1 + 0.09(C - 1)^{0.35}] \quad (7)$$

where $Z_{0(c=1)}$ is the thickness of the compressive stress layer when $C=1$.

Calculation of Tensile Residual Stress

It is very difficult to determine the tensile residual stress experimentally or analytically. On the basis of elastic-plastic finite element analyses, we have obtained a simulating equation expressing the distribution regularity of tensile

residual stress[3]:

$$\sigma_t(Z) = (Z - Z_0)^{1.35} / (a(Z - Z_0)^2 + b) \quad (8)$$

where a and b are two constants which can be determined from following conditions:

$$\int_0^{Z_0} \sigma_t(Z) dZ = \int_{Z_0}^{h/2} \sigma_t(Z) dZ \quad (9)$$

and

$$(Z_{tw} - Z_0) / Z_0 = 0.23 \quad (10)$$

Using Eqs.(8) and (10), we can obtain the value of σ_{tw} and Z_{tw} in Eqs.(1-1) and (2).

Determination of Internal Fatigue strength (IFS) of Target Material

IFS can be calculated from fatigue test data of one or two groups of peened and then ground specimen. Since IFS is an intrinsic parameter of property of given metal, it can be used to predict the FS of specimens peened under different conditions. But it has been established that the ratio of IFS and SFS, α , is approximately equal to 1.35. On the basis of theoretical analyses, it is believed that the fact $\alpha \approx 1.35$ is valid not only for the material used in Ref[4], but also for other materials. Then, a more convenient method to determine IFS may be used, i.e., to calculate IFS according to

$$IFS = \alpha \cdot SFS \quad (11)$$

where $\alpha = 1.35$.

By using Eqs(1-1),(8),(10) and (11), all main parameters needed can be determined and, then, the nominal fatigue strength (FS) of peened and then ground specimens can be calculated according to Eq.(2).

EXPERIMENTAL VERIFICATION

The specimens used this work were made of 40Cr steel, the mechanical properties of which after quenching and tempering at 200°C (A) and 550°C (C) are given in Tab.1. Fatigue specimens of 10×15×50mm were peened and then ground, and then tested under three-point bending condition with load ratio of 0.05 to obtain (nominal) fatigue strength (FS) for 5×10⁶ cycles life. A tentative peening procedure was carried out to determine D₄ in order to calculate equivalent load F during peening. The peening conditions and D₄ are also given in Tab.1.

The FS of as-heat-treated and un-peened specimens (i.e. the SFS of target material) and those of peened and then ground specimens are listed in Tab.2 and Tab.3. The parameters needed for prediction of FS were determined according to above proposed method and are also listed in Tab.2 and Tab.3. Finally, in the last column of Tab. 2 and Tab. 3, the predicted data of FS of specimens are given. It can be seen from comparison of experimental and calculated FS data that nearly all predicted data are

Tab.1 Mechanical Properties and Peening Conditions of Specimens

Symbol of specimen	Tempering temperature	σ_s , MPa	σ_b , MPa	D, mm	Air pressure, MPa	Coverage rate, %	Ground thickness, μm	D_d , μm
A000*								
A523				0.55	0.2	300	0	153
A121P20	200°C	1420	1910	1.10	0.2	100	20	271
A143P50				1.10	0.4	300	50	310
A166P50				1.10	0.6	600	50	355
C000*								
C523P20	550°C	980	1120	0.55	0.2	300	20	195
C143P50				1.10	0.4	300	50	353

* as un-peened

very near to, but a little lower than the experimentally determined ones. The results show clearly that the prediction method proposed in this work is successful.

Tab. 2 Experimental and Predicted Data of Nominal Fatigue Strength of Peened and Then Ground Specimens(the compressive residual stress field is determined by using the empirical method)

Symbol of specimen	Z _o , μ m	Z _s [*] μ m	a × 10 ⁻⁴	b	Z _{tm} , μ m	σ _{tm} MPa	IFS ^{**} , MPa	FS, MPa	
								Experimental	Predicted
A000		0						1060 ⁺	
A523	177	240	2.98	0.27	230	196	1430	1320	1300
A121P20	258	360	1.92	0.40	352	231		1340	1290
A143P50	330	470	1.68	0.47	406	241		1350	1300
A166P50	418	620	1.22	0.59	538	276		1360	1300
C000		0						820 ⁺	
C523P20	237	275	3.71	0.52	288	136	1110	1020	1030
C143P50	400	450	2.10	0.85	491	170		1040	1040

* measured depth of fatigue source

** calculated by using $IFS = 1.35 \times SFS$

+ SFS

Tab. 3 Experimental and preuted Data of Nominal Fatigue Strength of Peened and Then Ground Specimens(the compressive residual stress field is determined by using the analytical method)

Symbol of specimen	Z _o , μ m	Z _s [*] μ m	a × 10 ⁻⁴	b	Z _{tm} , μ m	σ _{tm} MPa	IFS ^{**} , MPa	FS, MPa	
								Experimental	Predicted
A000		0						1060 ⁺	
A523	181	240	3.10	0.26	222	192	1430	1320	1300
A121P20	270	360	2.02	0.38	332	228		1340	1290
A143P50	334	470	1.69	0.48	410	236		1350	1300
A166P50	438	620	1.26	0.61	538	267		1360	1310
C000		0						820 ⁺	
C523P20	240	275	3.69	0.54	295	135	1110	1020	1030
C143P50	420	450	1.85	0.83	517	187		1040	1030

* measured depth of fatigue source

** calculated by using IFS = 1.35 × SFS; + SFS

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